

AEROSPACE MATERIAL SPECIFICATION

SAE AMS4221

REV. D

Issued 1988-07
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Superseding AMS4221C

Aluminum Alloy, Plate
4.4Cu - 1.5Mg - 0.60Mn (2124-T8151)
Solution Heat Treated, Stress Relieved, and Precipitation Heat Treated
(Composition similar to UNS A92124)

RATIONALE

AMS4221D removes the alternate test for demonstration of fracture toughness (prior paragraph 3.3.4) and periodic tests, revises acceptance tests (4.2.1), and reports (4.4), and results from a Five Year Review and update of this specification.

1. SCOPE

1.1 Form

This specification covers an aluminum alloy in the form of plate 1.5 inches (38 mm) and over in nominal thickness.

1.2 Applications

This plate has been used typically for parts requiring a high level of mechanical properties up to 300 °F (149 °C), but usage is not limited to such applications.

1.2.1 Certain design and fabrication procedures may cause plate to become susceptible to stress-corrosion cracking; ARP823 recommends practices to minimize such conditions.

2. APPLICABLE DOCUMENTS

The issue of the following documents in effect on the date of the purchase order forms a part of this specification to the extent specified herein. The supplier may work to a subsequent revision of a document unless a specific document issue is specified. When the referenced document has been cancelled and no superseding document has been specified, the last published issue of that document shall apply.

2.1 SAE Publications

Available from SAE International, 400 Commonwealth Drive, Warrendale, PA 15096-0001, Tel: 877-606-7323 (inside USA and Canada) or 724-776-4970 (outside USA), www.sae.org.

AMS2355 Quality Assurance, Sampling and Testing, Aluminum Alloys and Magnesium Alloy, Wrought Products (Except Forging Stock), and Rolled, Forged, or Flash Welded Rings

AMS2772 Heat Treatment of Aluminum Alloy Raw Materials

ARP823 Minimizing Stress-Corrosion Cracking in Wrought Heat-Treatable Aluminum Alloy Products

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<http://www.sae.org/technical/standards/AMS4221D>**

2.2 ASTM Publications

Available from ASTM International, 100 Barr Harbor Drive, P.O. Box C700, West Conshohocken, PA 19428-2959, Tel: 610-832-9585, www.astm.org.

ASTM B 594	Ultrasonic Examination of Aluminum-Alloy Wrought Products for Aerospace Applications
ASTM B 645	Linear-Elastic Plane-Strain Fracture Toughness Testing of Aluminum Alloys
ASTM B 660	Packaging/Packing of Aluminum and Magnesium Products
ASTM B 666/B 666M	Identification Marking of Aluminum Products
ASTM E 399	Linear-Elastic Plane-Strain Fracture Toughness (K_{Ic}) of Metallic Materials
ASTM G 47	Standard Test Method for Determining Susceptibility to Stress-Corrosion Cracking of 2XXX and 7XXX Aluminum Alloy Products

2.3 ANSI Publications

Available from American National Standards Institute, 25 West 43rd Street, New York, NY 10036-8002, Tel: 212-642-4900, www.ansi.org.

ANSI H35.2	Dimensional Tolerances for Aluminum Mill Products
ANSI H35.2M	Dimensional Tolerances for Aluminum Mill Products (Metric)

3. TECHNICAL REQUIREMENTS

3.1 Composition

Shall conform to the percentages by weight shown in Table 1, determined in accordance with AMS2355.

TABLE 1 - COMPOSITION

Element	min	max
Silicon	--	0.20
Iron	--	0.30
Copper	3.8	4.9
Manganese	0.30	0.9
Magnesium	1.2	1.8
Chromium	--	0.10
Zinc	--	0.25
Titanium	--	0.15
Other Elements, each	--	0.05
Other Elements, total	--	0.15
Aluminum	remainder	

3.2 Condition

Solution heat treated, stress relieved by stretching to produce a nominal permanent set of 2% but not less than 1-1/2% nor more than 3%, and precipitation heat-treated; heat treatment shall be in accordance with AMS2772.

3.2.1 Plate shall receive no further straightening operations after stretching.

3.3 Properties

Plate shall conform to the following requirements in accordance with AMS2355 except that fracture toughness (3.3.3) shall be performed as indicated in the appropriate paragraph; tests are to be performed on the mill produced size.

3.3.1 Tensile Properties

Shall be as shown in Table 2, determined in accordance with AMS2355; tensile properties apply in the long-transverse direction and, when specified, in the longitudinal and/or short-transverse direction.

TABLE 2A - MINIMUM TENSILE PROPERTIES, INCH/POUND UNITS

Nominal Thickness Inches	Specimen Orientation	Tensile Strength ksi	Yield Strength at 0.2% Offset ksi	Elongation in 2 Inches or 4D %
1-1/2 to 2, incl	Longitudinal	68.0	62.0	8
	Long-Trans	68.0	62.0	7
	Short-Trans.	65.0	57.0	2
Over 2 to 3, incl	Longitudinal	67.0	60.0	8
	Long-Trans	67.0	60.0	7
	Short-Trans.	64.0	57.0	2
Over 3 to 4, incl	Longitudinal	67.0	59.0	7
	Long-Trans	67.0	59.0	6
	Short-Trans.	64.0	57.0	2
Over 4 to 5, incl	Longitudinal	66.0	58.0	6
	Long-Trans	66.0	58.0	6
	Short-Trans.	63.0	55.0	2
Over 5 to 6, incl	Longitudinal	65.0	57.0	6
	Long-Trans	65.0	56.0	5
	Short-Trans.	62.0	54.0	2

TABLE 2B - MINIMUM TENSILE PROPERTIES, SI UNITS

Nominal Thickness Millimeters	Specimen Orientation	Tensile Strength MPa	Yield Strength at 0.2% Offset MPa	Elongation in 50.8 mm or 4D %
38 to 51, incl	Longitudinal	469	427	8
	Long-Trans	469	427	7
	Short-Trans.	448	393	2
Over 51 to 76, incl	Longitudinal	462	414	8
	Long-Trans	462	414	7
	Short-Trans.	441	393	2
Over 76 to 102, incl	Longitudinal	462	407	7
	Long-Trans	462	407	6
	Short-Trans.	441	393	2
Over 102 to 127, incl	Longitudinal	455	400	6
	Long-Trans	455	400	6
	Short-Trans.	434	379	2
Over 127 to 152, incl	Longitudinal	448	393	6
	Long-Trans	448	386	5
	Short-Trans.	427	372	2

3.3.2 Stress-Corrosion Resistance

When specified, specimens as in 4.3.2 from plate 1.5 to 6 inches (38 to 152 mm), inclusive, in nominal thickness shall show no evidence of stress-corrosion cracking when stressed to 50% of the specified minimum long-transverse yield strength and tested in accordance with ASTM G 47.

3.3.3 Fracture Toughness

For product 1.500 to 6.000 inches (38.10 to 152.40) thick, plane-strain fracture toughness (K_{Ic}) shall be determined in accordance with ASTM E 399 and ASTM B 645 in the L-T specimen orientation, and other orientations as specified by the purchaser. Values shall conform to minimum values in Table 3.

TABLE 3A - MINIMUM FRACTURE TOUGHNESS PARAMETERS, INCH/POUND UNITS

Nominal Thickness Inches	Specimen Orientation	Specimen Orientation	Specimen Orientation
	L – T	T – L	S – L
	ksi√inch	ksi√inch	ksi√inch
1.500 to 2.000, incl	26	23	19
Over 2.000 to 4.000, incl	26	22	20
Over 4.000 to 6.000, incl	25	21	21

TABLE 3B - MINIMUM FRACTURE TOUGHNESS PARAMETERS, SI UNITS

Nominal Thickness Millimeters	Specimen Orientation	Specimen Orientation	Specimen Orientation
	L – T	T – L	S – L
	MPa√m	MPa√m	MPa√m
38.10 to 50.80, incl	28.6	25.3	20.9
Over 50.80 to 101.60, incl	28.6	24.2	22.0
Over 101.60 to 152.40, incl	27.5	23.1	23.1